

Put in $a = 2M_3 G/c^2$

$$\text{Then, } au = \frac{2M_3 G}{rc^2}$$

Consider the case of the earth. Assuming a circular orbit, we have

$$\frac{m_e v^2}{r} = \frac{m M_3 G}{r^2}$$

or
$$v^2 = \frac{M_3 G}{r}$$

$$\therefore au = 2 \frac{v^2}{c^2} = 2 \times \left(\frac{2\pi R_{\oplus}}{1 \text{ light year}} \right)^2$$

$$= 2 \times \left(\frac{2\pi \times .93 \times 10^8}{5.88 \times 10^{12}} \right)^2 \approx 2 \times 10^{-8}$$

$$au|_{\text{earth}} \approx 2 \times 10^{-8}, \quad au|_{\text{mercury}} \approx 5.1 \times 10^{-8}$$

Put in value for a in the orbit eqn \Rightarrow

$$\left(\frac{du}{d\phi} \right)^2 + u^2 = \frac{2E}{L^2} + \frac{2M_3 G}{L^2} u + \frac{2M_3 G}{c^2} u^3$$

Let us compare this with Newton's result:

Equation 1 of notes gives with $V(r) = -\frac{mM_3 G}{r}$
and $m=1$